# Numerical Simulation of Plane Crack Using Hermite Cubic Spline Wavelet

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**Abstract:** Two-dimensional wavelet-based numerical approximation using Hermite cubic spline wavelet on the interval (HCSWI) is proposed to solve stress intensity factors (SIFs) of plate structures. The good localization property of wavelets is used to approximate displacement fields by multi-scale bases of HCSWI. Example computations are performed for plates with a central crack and double edge cracks. The numerical results prove that, compared with the conventional finite element method and the analytical solutions, the new procedure are efficient in both its accuracy and its reduction of degree of freedoms (DOFs).

**Keywords:** Plate structures; Stress intensity factors; HCSWI; Wavelet numerical method

## 1 Introduction

Crack propagation behavior is a major issue in a variety of structures of industries. Aerospace structures, gas turbine blade, pressure vessels, pipelines, and wind turbine blades are obvious examples where failure could lead to catastrophic consequences and loss of life. The solution of stress intensity factors (SIFs) involves a well-known mathematical difficulty. Among calculation algorithms in problems of fracture mechanics, finite element analysis (FEA) is the most commonly used method [Tada et al. (2000)]. Recently, many numerical methods were proposed to efficiently calculate SIFs. Wearing et al. [Wearing and Ahmadi-Brooghani (1999)] presented a boundary element method to analyze two-dimensional crack problems. Giner et al. proposed an extended finite element method (X-FEM) to solve crack problems and further developed an implementation of the X-FEM into the commercial FEA software Abaqus [Giner et al. (2009)]. In the above mentioned methods,

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the efficiency and precision are key issues to overcome the large numerical error caused by crack singularity. Adaptive FEM is also an important technique to obtain more accurate values of SIFs. Kpegba et al. proposed a hybrid method to combine the two independent classical meshes to improve the precision of traditional FEA [Kpegba and Ottavy (1996)].

Wavelet numerical method was proposed to solve partial differential equations (PDEs) [Diaz et al. (2009); Vampa et al. (2010); Ma et al. (2003); Chen et al. (2009); Wang et al. (2010, 2011); He et al. (2007); Xiang et al. (2009a, 2007)]. Hermite cubic spline wavelet on the interval (HCSWI) is the new wavelet bases constructed by Jia and Liu [Jia and Liu (2006); Jia (2009)]. Xiang et al. employed HCSWI to analyze Possion equation [Xiang et al. (2009b)]. The advantage of HCSWI wavelet-based numerical method is the decoupling of multi-scaling approximation equations [Xiang et al. (2009b)].

In the present work, a new wavelet numerical method using HCSWI is proposed to calculate SIFs of plate structures. Two examples of plates with a central crack and double edge cracks are investigated. It notes that the extension of the present method to other classes of structures with cracks is possibly.

#### 2 The wavelet bases of HCSWI

In this section, we give a brief description of HCSWI. One-dimensional multiresolution analysis (MRA) of HCSWI is given by Jia and Liu [Jia and Liu (2006)]. The scaling functions  $\phi_{1,k}$  (scale=1 and k=1, 2, 3, 4) are defined by

$$\begin{cases} \phi_{1,1}(x) := \sqrt{\frac{5}{24}} \phi_1(2x-1) \\ \phi_{1,2}(x) := \sqrt{\frac{15}{4}} \phi_2(2x) \\ \phi_{1,3}(x) := \sqrt{\frac{15}{8}} \phi_2(2x-1) \\ \phi_{1,4}(x) := \sqrt{\frac{15}{4}} \phi_2(2x-2) \end{cases}$$
(1)

and the wavelets  $\psi_{j,k}$  (scale=*j* and *k*=1, 2, ..., 2<sup>*j*+1</sup>) are

$$\begin{cases} \psi_{j,k}(x) := \frac{2^{-j/2}}{\sqrt{729.6}} \psi_1\left(2^j x - \frac{k}{2}\right) & \text{for } k = 2, 4, \dots, 2^{j+1} - 2, \\ \psi_{j,k}(x) := \frac{2^{-j/2}}{\sqrt{153.6}} \psi_2\left(2^j x - \frac{k-1}{2}\right) & \text{for } k = 3, \dots, 2^{j+1} - 1, \\ \psi_{j,1}(x) := \frac{2^{-j/2}}{\sqrt{76.8}} \psi_2\left(2^j x\right), \\ \psi_{j,2^{j+1}}(x) := \frac{2^{-j/2}}{\sqrt{76.8}} \psi_2\left(2^j x - 2^j\right), \end{cases}$$

$$(2)$$

in which  $\phi_1$  and  $\phi_2$  be the cubic splines supported on the interval [-1, 1] as

$$\phi_1(x) := \begin{cases} (x+1)^2(1-2x) & \text{for } x \in [-1,0] \\ (1-x)^2(1+2x) & \text{for } x \in [0,1] \\ 0 & \text{for } x \notin [-1,1] \end{cases}$$

and

$$\phi_2(x) := \begin{cases} (x+1)^2 x & \text{for } x \in [-1,0] \\ (x-1)^2 x & \text{for } x \in [0,1] \\ 0 & \text{for } x \notin [-1,1] \end{cases}$$
(3)

and the wavelets  $\psi_1$  and  $\psi_2$  are supported on the interval [-1, 1] as

$$\begin{cases} \psi_1(x) = -2\phi_1(2x+1) + 4\phi_1(2x) - 2\phi_1(2x-1) - 21\phi_2(2x+1) + 21\phi_2(2x-1) \\ \psi_2(x) = \phi_1(2x+1) - \phi_1(2x-1) + 9\phi_1(2x+1) + 12\phi_2(2x) + 9\phi_2(2x-1) \end{cases}$$
(4)

Hermite cubic splines  $\phi_1$  and  $\phi_2$ , wavelets  $\psi_1$  and  $\psi_2$  are shown in Figs.1 and 2, respectively



The special property of HCSWI is

$$\begin{cases} \left\langle \phi_{1,k_{1}}^{\prime},\psi_{j,k_{2}}^{\prime}\right\rangle = \int_{0}^{1}\phi_{1,k_{1}}^{\prime}\psi_{j,k_{2}}^{\prime}dx = 0 \quad \text{for all } j \\ \left\langle \psi_{j_{1},k_{1}}^{\prime},\psi_{j_{2},k_{2}}^{\prime}\right\rangle = \int_{0}^{1}\psi_{j_{1},k_{1}}^{\prime}\psi_{j_{2},k_{2}}^{\prime}dx = 0 \quad \text{for } j_{1} \neq j_{2} \end{cases}$$
(5)



Figure 2: Wavelets  $\psi_1$  and  $\psi_2$ 

The wavelet bases in scale space  $V_i$  can be written by

$$\boldsymbol{\phi}_j = [\boldsymbol{\phi}_1, \boldsymbol{\psi}_1, \boldsymbol{\psi}_2, \cdots, \boldsymbol{\psi}_{j-1}] \tag{6}$$

where  $\boldsymbol{\phi}_1 = [\phi_{1,1}, \phi_{1,2}, \phi_{1,3}, \phi_{1,4}]$  denotes scaling functions in  $V_1$ , and  $\boldsymbol{\psi}_s(s = 1, 2, \dots, j - 1)$  consists of the wavelet bases in wavelet space  $W_s$ , i.e.,  $\boldsymbol{\psi}_s = [\psi_{s,1}, \psi_{s,2}, \dots, \psi_{s,2^{s+1}}]$ . Fig. 3 shows the first derivative of  $\boldsymbol{\phi}_1$  and  $\boldsymbol{\psi}_1$ .



To construct two-dimensional wavelet bases, tensor product of one-dimensional wavelets is a direct way [Chen et al. (2004); Zhang et al. (2010); Mallat (1999)]. Take  $\phi_2 = [\phi_1, \psi_1]$  for example, the two-dimensional scaling and wavelet functions

are

$$\boldsymbol{\phi}_2 \otimes \boldsymbol{\phi}_2 = [\boldsymbol{\phi}_1 \otimes \boldsymbol{\phi}_1, \boldsymbol{\phi}_1 \otimes \boldsymbol{\psi}_1, \boldsymbol{\psi}_1 \otimes \boldsymbol{\phi}_1, \boldsymbol{\psi}_1 \otimes \boldsymbol{\psi}_1] = [\boldsymbol{\phi}^1, \boldsymbol{\psi}^2, \boldsymbol{\phi}^3, \boldsymbol{\psi}^4]$$
(7)

where  $\otimes$  is the kronecker symbol. Therefore, we obtain four functions, namely, the scaling functions  $\phi^1 = \phi_1 \otimes \phi_1$ , wavelets  $\psi^1 = \psi_1 \otimes \phi_1$ ,  $\psi^2 = \phi_1 \otimes \psi_1$  and  $\psi^3 = \psi_1 \otimes \psi_1$ , which are shown in Fig.4(a), (b), (c) and (d), respectively.



Figure 4: Two-dimensional HCSWI scaling functions  $\phi^1$  and wavelets  $\psi^1$ ,  $\psi^2$  and  $\psi^3$ 

#### 3 Numerical computation formulas using HCSWI

Plane elasticity includes plane stress and plane strain problems. Here, the plane stress problem is analyzed and the plane strain problem is similarly if Young's

modulus *E* and Poisson's ratio  $\mu$  are replaced by  $E/(1-\mu^2)$  and  $\mu/(1-\mu)$  respectively. The generalized function of potential energy for plane stress problem is [Zienkiewicz, O.C., Taylor, R.L. (2000)]

$$\mathbf{\Pi}_{p}(\boldsymbol{\delta}) = \int_{\Omega} \frac{1}{2} \boldsymbol{\varepsilon}^{\mathrm{T}} \mathbf{D} \boldsymbol{\varepsilon} t dx dy - \int_{\Omega} \boldsymbol{\delta}^{\mathrm{T}} \mathbf{f} t dx dy - \int_{S_{\sigma}} \boldsymbol{\delta}^{\mathrm{T}} \mathbf{p} t ds - \sum_{i=1}^{n} \boldsymbol{\delta}_{i}^{\mathrm{T}} \mathbf{F}_{i}$$
(8)

where  $\Omega$  is the solving domain with length  $l_x$  and  $l_y,t$  is the plate thickness,  $\mathbf{f} = \{f_x f_y\}^T$  are the body forces,  $\boldsymbol{\delta} = \{u v\}^T$  are the displacements,  $\mathbf{p} = \{p_x p_y\}^T$  are the surface forces,  $\mathbf{F}_i = \{F_{xi} F_{yi}\}^T$  are point forces,  $\boldsymbol{\delta}_i = \{u_i v_i\}^T$  are the point displacements,  $\mathbf{D}$  is the elastic matrix, and  $\boldsymbol{\varepsilon} = \{\varepsilon_x \varepsilon_y \gamma_{xy}\}^T$  are the strain components. **D** and  $\boldsymbol{\varepsilon}$  are given by

$$\mathbf{D} = \frac{E}{1 - \mu^2} \begin{bmatrix} 1 & \mu & 0\\ \mu & 1 & 0\\ 0 & 0 & \frac{1 - \mu}{2} \end{bmatrix}$$

and

**–** )

$$\boldsymbol{\varepsilon} = \begin{bmatrix} \frac{\partial}{\partial x} & 0\\ 0 & \frac{\partial}{\partial y}\\ \frac{\partial}{\partial y} & \frac{\partial}{\partial x} \end{bmatrix} \mathbf{u}$$
(9)

The relationship between stress and strain is

$$\boldsymbol{\sigma} = \{\boldsymbol{\sigma}_x \ \boldsymbol{\sigma}_y \ \boldsymbol{\tau}_{xy}\}^T = \mathbf{D}\boldsymbol{\varepsilon}$$
(10)

The two-dimensional HCSWI bases are employed as interpolating functions to construct multi-scale approximation equations as

$$u = \boldsymbol{\phi}_j^{\mathrm{T}} \mathbf{u}$$

and

$$\boldsymbol{v} = \boldsymbol{\phi}_j^{\mathrm{T}} \mathbf{v} \tag{11}$$

where

$$\begin{cases} \mathbf{u} = \{u_1 \dots u_{2^{j+1}} | \cdots | u_{2^{j+2}-2^{j+1}+1} \dots u_{2^{j+1}}\}^{\mathrm{T}} \\ \mathbf{v} = \{v_1 \dots v_{2^{j+1}} | \cdots | v_{2^{j+2}-2^{j+1}+1} \dots v_{2^{j+1}}\}^{\mathrm{T}} \end{cases}$$
(12)

are the column vectors of wavelet interpolating coefficients ( also the DOFs) to be determined.

$$\begin{bmatrix} \mathbf{K}^1 & \mathbf{K}^2 \\ \mathbf{K}^3 & \mathbf{K}^4 \end{bmatrix} \begin{bmatrix} \mathbf{u} \\ \mathbf{v} \end{bmatrix} = \begin{bmatrix} \mathbf{P}_a \\ \mathbf{P}_b \end{bmatrix}$$
(13)

where

$$\mathbf{P}_{a} = \int_{S_{\sigma}} p_{x} \boldsymbol{\phi}_{j}^{\mathrm{T}} \mathrm{d}s + \int_{\Omega} f_{x} \boldsymbol{\phi}_{j}^{\mathrm{T}} \mathrm{d}\Omega + \sum_{i=1}^{n} \boldsymbol{\phi}_{j}^{\mathrm{T}} (\boldsymbol{\xi}_{i}, \boldsymbol{\eta}_{i}) F_{ix} \}$$
(14)

$$\mathbf{P}_{b} = \int_{S_{\sigma}} p_{y} \boldsymbol{\phi}_{j}^{\mathrm{T}} \mathrm{d}s + \int_{\Omega} f_{y} \boldsymbol{\phi}_{j}^{\mathrm{T}} \mathrm{d}\Omega + \sum_{i=1}^{n} \boldsymbol{\phi}_{j}^{\mathrm{T}} (\xi_{i}, \eta_{i}) F_{iy}$$
(15)

$$\mathbf{K}^{1} = E/(1-\mu^{2})(\mathbf{A}_{1}^{11} \otimes \mathbf{A}_{2}^{00} + (1-\mu)/2\mathbf{A}_{1}^{00} \otimes \mathbf{A}_{2}^{11})$$
(16)

$$\mathbf{K}^{2} = E/(1-\mu^{2})(\mu \mathbf{A}_{1}^{10} \otimes \mathbf{A}_{2}^{01} + (1-\mu)/2\mathbf{A}_{1}^{01} \otimes \mathbf{A}_{2}^{10})$$
(17)

$$\mathbf{K}^{3} = (\mathbf{K}^{2})^{1}$$
(18)  
$$\mathbf{K}^{4} = E/(1-\mu^{2})(\mathbf{A}_{1}^{00} \otimes \mathbf{A}_{2}^{11} + (1-\mu)/2\mathbf{A}_{1}^{11} \otimes \mathbf{A}_{2}^{00})$$
(19)

$$\mathbf{K}^{4} = E/(1-\mu^{2})(\mathbf{A}_{1}^{00} \otimes \mathbf{A}_{2}^{11} + (1-\mu)/2\mathbf{A}_{1}^{11} \otimes \mathbf{A}_{2}^{00})$$
(19)  
in which

in which

$$\mathbf{A}_{1}^{00} = l_{x} \int_{0}^{1} \boldsymbol{\phi}_{j}^{\mathrm{T}} \boldsymbol{\phi}_{j} \mathrm{d}\boldsymbol{\xi}$$
<sup>(20)</sup>

$$\mathbf{A}_{1}^{01} = \int_{0}^{1} \boldsymbol{\phi}_{j}^{T} \frac{\mathrm{d}\boldsymbol{\phi}_{j}}{\mathrm{d}\boldsymbol{\xi}} \mathrm{d\boldsymbol{\xi}}$$
(21)

$$\mathbf{A}_{1}^{10} = (\mathbf{A}_{1}^{01})^{\mathrm{T}}$$
(22)

$$\mathbf{A}_{1}^{11} = 1/l_{x} \int_{0}^{1} \frac{\mathrm{d}\boldsymbol{\phi}_{j}}{\mathrm{d}\boldsymbol{\xi}} \frac{\mathrm{d}\boldsymbol{\phi}_{j}}{\mathrm{d}\boldsymbol{\xi}} \mathrm{d}\boldsymbol{\xi}$$
(23)

 $\mathbf{A}_{2}^{lm}(l,m=0,1)$  is similarly to  $\mathbf{A}_{1}^{lm}(l,m=0,1)$  when  $l_{x}$  and  $d\xi$  are replaced by  $l_{y}$  and  $d\eta$  respectively.

The lifting scheme for multi-scale calculating SIFs is similar to Ref.[Xiang et al. (2009b)]. Fig.5 shows the lifting scheme to lift the scale from 1 to j.

When the scale 1 is lifted to scale *j*,  $\mathbf{A}_1^{11}$  and  $\mathbf{A}_2^{11}$  would be decomposed totally across different scale. The integral  $\mathbf{A}_1^{11}$  is

$$\mathbf{A}_{1}^{11} = 1/l_{x} \begin{bmatrix} \int_{0}^{1} \frac{d\boldsymbol{\phi}_{1}^{T}}{d\xi} \frac{d\boldsymbol{\phi}_{1}}{d\xi} d\boldsymbol{\xi} & \mathbf{0} & \cdots & \mathbf{0} \\ & \int_{0}^{1} \frac{d\boldsymbol{\psi}_{1}^{T}}{d\xi} \frac{d\boldsymbol{\psi}_{1}}{d\xi} d\boldsymbol{\xi} & \cdots & \mathbf{0} \\ \text{symmetry} & \ddots & \vdots \\ & & & \int_{0}^{1} \frac{d\boldsymbol{\psi}_{j-1}^{T}}{d\xi} \frac{d\boldsymbol{\psi}_{j-1}}{d\xi} d\boldsymbol{\xi} \end{bmatrix}$$
(24)



Figure 5: The lifting scheme of wavelet bases

The integrals  $\mathbf{A}_1^{01}$ ,  $\mathbf{A}_2^{01}$ ,  $\mathbf{A}_1^{00}$  and  $\mathbf{A}_2^{00}$  would be preserved coupling relationship. However, the former generated sub-matrices need not be re-calculated. This property can also increase the calculating efficiency. The integral  $\mathbf{A}_1^{01}$  is

$$\mathbf{A}_{1}^{01} = \begin{bmatrix} \int_{0}^{1} \boldsymbol{\phi}_{1}^{T} \frac{d\boldsymbol{\phi}_{1}}{d\xi} d\xi & \int_{0}^{1} \boldsymbol{\phi}_{1}^{T} \frac{d\boldsymbol{\psi}_{1}}{d\xi} d\xi & \cdots & \int_{0}^{1} \boldsymbol{\phi}_{1}^{T} \frac{d\boldsymbol{\psi}_{j-1}}{d\xi} d\xi \\ & \int_{0}^{1} \boldsymbol{\psi}_{1}^{T} \frac{d\boldsymbol{\psi}_{1}}{d\xi} d\xi & \cdots & \int_{0}^{1} \boldsymbol{\psi}_{1}^{T} \frac{d\boldsymbol{\psi}_{j-1}}{d\xi} d\xi \\ \text{symmetry} & \ddots & \vdots \\ & & & \int_{0}^{1} \boldsymbol{\phi}_{j-1}^{T} \frac{d\boldsymbol{\psi}_{j-1}}{d\xi} d\xi \end{bmatrix}$$
(25)

and the integral  $\mathbf{A}_1^{00}$  is

$$\mathbf{A}_{1}^{00} = l_{x} \begin{bmatrix} \int_{0}^{1} \boldsymbol{\phi}_{1}^{T} \boldsymbol{\phi}_{1} \mathrm{d}\boldsymbol{\xi} & \int_{0}^{1} \boldsymbol{\phi}_{1}^{T} \boldsymbol{\psi}_{1} \mathrm{d}\boldsymbol{\xi} & \cdots & \int_{0}^{1} \boldsymbol{\phi}_{1}^{T} \boldsymbol{\psi}_{j-1} \mathrm{d}\boldsymbol{\xi} \\ & \int_{0}^{1} \boldsymbol{\psi}_{1}^{T} \boldsymbol{\phi}_{1} \mathrm{d}\boldsymbol{\xi} & \cdots & \int_{0}^{1} \boldsymbol{\psi}_{1}^{T} \boldsymbol{\psi}_{j-1} \mathrm{d}\boldsymbol{\xi} \\ \text{symmetry} & \ddots & \vdots \\ & & & \int_{0}^{1} \boldsymbol{\phi}_{j-1}^{T} \boldsymbol{\psi}_{j-1} \mathrm{d}\boldsymbol{\xi} \end{bmatrix}$$
(26)

#### 4 Numerical investigations

In this study, the displacement extrapolation technique has been used to calculate the SIFs as follows [Tada et al. (2000)]

$$v = \frac{K_I}{2G} \sqrt{\frac{r}{2\pi}} (k+1) \tag{27}$$

where *r* is the distance from crack tip to a point considered along with the crack edge,  $k = (3 - \mu)/(1 + \mu)$  is the elastic parameter,  $G = \frac{E}{2(1 + \mu)}$  is shear modulus. Therefore, we obtain SIFs for crack with mode I as

$$K_I = \frac{2Gv}{(k+1)} \sqrt{\frac{2\pi}{r}}$$
(28)



Figure 6: The center crack problem and numerically computational model

#### 4.1 Example 1: The center crack problem

The central cracked plate and the computational model are shown in Fig.6(a) and(b) respectively. The geometry is imposed by a plane stress condition with symmetrical uniform load p applied under mode I loading condition. Plate length is 2b, width L and centre crack width is 2a. The analytical SIFs for this problem is given by Ref. [Tada et al. (2000)] as

$$K_I = p\sqrt{\pi a}F(a/b) \tag{29}$$

To make a simple comparison, the non-dimensional stress intensity factor is defined as follows

$$Z_I = \frac{K_I}{p\sqrt{a}} \tag{30}$$

Error estimate is necessary to make an adaptive wavelet numerical calculation. If the SIFs of wavelet numerical solution  $Z_I^{i+1}$  and  $Z_I^i$  (*i*+1 and *i* denote the neighbor scales) is chosen as the benchmark, with the defined dimensionless relative error (error estimate) as the following equation

$$\boldsymbol{\varepsilon}^* = \left| \boldsymbol{Z}^{i+1} - \boldsymbol{Z}^i \right| / \boldsymbol{Z}^{i+1} \tag{31}$$

Given the error threshold  $\varepsilon_0^*$ , error estimate  $\varepsilon^*$  can be made according to Eq. (31). When an error  $\varepsilon^*$  in a certain scale is large than the error threshold  $\varepsilon_0^*$ , i.e.,  $\varepsilon^* > \varepsilon_0^*$ , scale will be lifted to the higher one. This process will be repeated until solution is less than error threshold  $\varepsilon_0^*$ . Therefore, the adaptive analysis algorithm is listed below:

(1) Give the error threshold  $\varepsilon_0^*$ 

(2) Use scaling functions scale  $\phi_i$  and  $\Phi_{i+1}$  to calculate the corresponding  $Z^i$  and  $Z^{i+1}$ , respectively.

(3) Calculate error estimate  $\varepsilon^*$  according to Eq. (29).

(4) Compare  $\varepsilon^*$  and  $\varepsilon_0^*$ , if  $\varepsilon_i^* > \varepsilon_0^*$ , then go back to step (2). Otherwise, stop the calculation and obtain the results.

Suppose the error threshold  $\varepsilon_0^* = 0.01$ , it found that only three interactions were performed to obtain a comfortable result at scale *j*=4, as shown in Table 1. Different methods are used to make a comparison, such as:

Method 1: Analytical solution, the value of F(a/b) is shown in Ref.[Tada et al. (2000)]. Method 2: Analytical solution, the other value of F(a/b) is calculated by [Tada et al. (2000)]

$$F(a/b) = \{1 - 0.025(\frac{a}{b})^2 + 0.06(\frac{a}{b})^4\} \sqrt{\sec\frac{\pi a}{2b}}$$
(32)

Method 3: Adaptive finite element method with about 5000 DOFs as shown in Ref. [Souiyah et al. (2009)].

Present method: The present results computed using HCSWI bases at level j=4. It notes that the DOFs of multi-scale approximation equation at each scale j=1, 2, 3, 4 are 32, 128, 512 and 2048, respectively.

The present results computed using HCSWI bases at level j=4 are found to be in good agreement with those in literature, as shown in Table 1. By comparing with other methods, the relative errors to the analysis solution is less than 0.261% for all values, which gives highly accurate results and much better than 2.195% of adaptive finite element method with about 5000 DOFs. In order to show the performance of the present method, we give a comparison of the computational cost of the present method with finite element method. All the computations are conducted using Matlab2010a on a laptop computer with a 2 GHz dual-core CPU (T4200) and 2GB memory. According to the TIC and TOC commands of Matlab2010a, the computing times required to Method 3 is about 4.5 seconds, whereas the present method is less than 3 seconds (only measured time for the level j=4). Moreover, the DOFs of adaptive finite element method are almost 2.5 times of those of the present method. Therefore, the performance of the present wavelet-based numeri-

-11-	Meth	nod 1		Method 2			Method 3		Present	method
ain	F(a/b)	$Z_I$	F(a/b)	$Z_I$	Error/%	F(a/b)	$Z_I$	Error/%	$Z_I^4$	Error/%
0.1	1.006	1.7831	1.006	1.783	0.006	1.001	1.7742	0.499	1.7833	0.011
0.2	1.0246	1.8161	1.0245	1.8158	0.017	1.0237	1.8145	0.088	1.8148	0.072
0.3	1.0557	1.8747	1.0575	1.8744	0.016	1.0416	1.8462	1.520	1.8698	0.261
0.4	1.1094	1.9664	1.109	1.9657	0.036	1.109	1.9657	0.036	1.9655	0.046
0.5	1.1867	2.1034	1.1862	2.1025	0.043	1.1879	2.1055	0.100	2.1031	0.014
0.6	1.3033	2.31	1.3028	2.3091	0.039	1.2747	2.2593	2.195	2.311	0.043
0.7	1.4882	2.6378	1.4874	2.6363	0.057	1.4655	2.5975	1.528	2.6352	0.099
0.8	1.816	3.2188	1.8143	3.2158	0.093	1.8131	3.2136	0.162	3.2253	0.202
0.9	2.5776	4.5687	2.5767	4.567	0.037	2.565	4.5463	0.490	4.5608	0.173

for different a/h (I/2h-1) for ev Ń 2 Table 1. Stress intensity facto

cal is verified and the good performance of the lifting scheme is observed when the wavelets are added step by step to realize multi-scale approximation of SIFs. The SIFs  $Z_I^1$ ,  $Z_I^2$  and  $Z_I^3$  and the corresponding  $\varepsilon^*$  of HCSWI wavelet bases at scale j=1, 2, 3 are listed in Table 2.

a/b	$Z_I^1$	$\boldsymbol{\varepsilon}^{*}$	$Z_I^2$	$\boldsymbol{\varepsilon}^{*}$	$Z_I^3$	$\boldsymbol{\varepsilon}^*$
0.1	1.5821	0.058	1.6793	0.050	1.7685	0.008
0.2	1.6944	0.022	1.7319	0.013	1.7555	0.033
0.3	1.8020	0.017	1.8326	0.022	1.8736	0.002
0.4	2.1053	0.047	2.0117	0.013	1.9854	0.010
0.5	2.2261	0.016	2.1910	0.031	2.1245	0.010
0.6	2.4758	0.024	2.4189	0.037	2.3327	0.009
0.7	2.7845	0.032	2.6980	0.028	2.6246	0.004
0.8	3.3725	0.015	3.3212	0.022	3.2488	0.007
0.9	4.4287	0.015	4.4956	0.011	4.5463	0.003

Table 2: Stress intensity factors  $Z_I^i$  (i=1,2,3) for different *a/b* (*L/2b*=1)

### 4.2 Example 2: The double edge crack problem

Figure 7 shows a double edge cracked plate and its computational model. We compare the present method using HCSWI bases at level j=4 with traditional finite element method (PLANE42 element in software ANSYS) with  $200 \times 200$ meshes (80000 DOFs). The benchmark to compare the two methods is the analytical solution given by [Tada et al. (2000)]

$$F(a/b) = \frac{1.122 - 0.561(\frac{a}{b}) - 0.205(\frac{a}{b})^2 + 0.471(\frac{a}{b})^3 - 0.190(\frac{a}{b})^4}{\sqrt{1 - a/b}}$$
(33)

Table 3 shows the relative errors between the two methods and the analytical solutions. The relative errors between results of the proposed method and the analytical solutions are less than 1.763% for all a/b, which are much smaller than that (5.926%) obtained using the traditional finite element method (PLANE42 element in software ANSYS) with 80000 DOFs.

By summarizing the above SIFs analysis and comparisons, the validity of the present method is testified.

### 5 Conclusions

Properties of wavelet of good localization are used to approximate displacement fields near the crack tip. Wavelet-based numerical method to analyze plate struc-





(a) The double edge crack (b) Computational Model

Figure 7:	The double-edge cracked	plate and it	s computational	model
U	U	1	1	

	Analytical solution		PLANE42		Proposed method	
aib	[Tada et a	et al. (2000)]				
	F(a/b)	$Z_I$	$Z_I$	Error/%	$Z_I$	Error/%
0.1	1.1219	1.9885	1.8860	5.437	1.9785	0.503
0.2	1.1237	1.9917	1.9633	1.446	1.9856	0.306
0.3	1.1312	2.005	2.0552	2.443	2.0212	0.808
0.4	1.1491	2.0367	2.1471	5.143	2.0726	1.763
0.5	1.1841	2.0987	2.2309	5.926	2.105	0.300
0.6	1.2471	2.2104	2.3181	4.646	2.245	1.565
0.7	1.3598	2.4102	2.4528	1.736	2.3875	0.942
0.8	1.5771	2.7954	2.7416	1.961	2.7784	0.608
0.9	2.1179	3.7539	3.5608	5.424	3.6932	1.617

Table 3: Stress intensity factors  $Z_I$  for different *a/b* under uniform tension (*L/2b*=1) for exmaple 2

tures using HCSWI bases is established to simulate singularity problems. For the good characteristics of the wavelet bases of HCSWI, such as multi-resolution analysis and orthogonal according to the inner product of  $\langle u', v' \rangle$ , the lifting scheme of the present method can be realized efficiently. Therefore, the present method has at least two advantages. The first advantage is that the computation efficiency,

necessary for achieving a solution with the same accuracy compared with the usual finite element method, are much reduced. The second advantage stems from the fact that the so-called multi-resolutions of wavelets make it possible to generate a multi-scale approximation equation for the singularity problems. For the orthogonal characteristic of the wavelet bases with respect to the given inner product, the corresponding multi-scale equations will be decoupled across scales partially and suit for nesting approximation. The numerical results show that wavelet numerical method is suitable to compute SIFs for plate structures.

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