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MHD MAXWELL FLUID FLOW IN PRESENCE OF NANO-PARTICLE THROUGH A VERTICAL POROUS-PLATE WITH HEAT-GENERATION, RADIATION ABSORPTION AND CHEMICAL REACTION

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ABSTRACT

Present study concerns with the numerical investigation of MHD transient naturally convective and higher order chemically reactive Maxwell fluid with Nano-particle flow through a vertical porous plate with the effects of heat generation and radiation absorption. A boundary layer approximation is carried out to develop a flow model representing time dependent momentum, energy, and concentration equations. The governing model equations in partial differential equations (PDEs) form are transformed into a set of nonlinear ordinary differential equation (ODEs) by using non-similar technique. Explicit Finite Difference Method (EFDM) is employed by implementing an algorithm in Compaq Visual Fortran 6.6a to solve the obtained set of nonlinear coupled ODEs. For optimizing the system parameter and accuracy of the system, the stability and convergence analysis (SCA) are carried out. It is observed that with initial boundary conditions, U = V = T = C = 0 and for $\Delta \tau = 0.005$, $\Delta X = 0.20$ and $\Delta Y = 0.25$, the system converged at Prandtl number, $P_r \ge 0.209$ and Lewis number, $L_e \ge 0.16$. The velocity, temperature and concentration flow are investigated and shown graphically with the effect of system parameters. Furthermore, the effect of system parameters on skin friction coefficient, C_f , Nusselt number, N_u , and Sherwood number, S_h , are also examined and tabularized.

Keywords: Maxwell fluid, Nano-particle, MHD, thermal radiation and radiation absorption.

1. INTRODUCTION

In the last few decades, the study of magnetohydrodynamic (MHD) heat-mass transfer flow finds vast significant importance in engineering technology industries. A wide range of applications can be found in several fields such as aerodynamic extrusion of plastic sheets, energy storage units, biological transportation, liquid metal fluids oil reservoirs, high-temperature plasmas, geothermal systems, heat insulation and metal and polymer extrusion, thermal energy storage devices, electronic cooling, boilers, nuclear process, micro MHD pumps, ground water systems etc. Investigations of MHD flows are mostly in vertical moving porous plate (Mohamed and Abo-Dahab, 2009), vertical porous plate (Islam et al. 2015; Khan et al. 2014; Murthy et al. 2015; Ramachandra et al. 2011), vertical insulated porous plate (Baoku et al. 2013), infinite inclined porous plate (Das et al. 2015), semi-infinite vertical porous plate (Ravikumar et al. 2014) etc.

In recent days, nanotechnology has received a lot of attention (Bég et al. 2014; Dogonchi and Ganji, 2016; Ferdows et al. 2013; Ibáñez et al. 2016; Kandasamy et al. 2016; Mosayebidorcheh et al. 2016; Pandey et al. 2016; Shahmohamadi and Rashidi, 2016; Sheikholeslami et al., 2016; Biswas et al, 2017; Arifuzzaman et al. 2017; Biswas et al, 2018) where the further development of higher performance is still going due to effective applications in the field of cooling (transformer cooling, electronics device cooling, silicon mirror cooling, vehicles cooling, controlling fusion), biomedical (magnetic cell separation, drug delivery, cancer therapeutics, cryopreservation, nanocryosurgery) etc. The term "nanofluid" can be refers to a class of fluids by suspending nanometre sized (1-100 nm diameters) particles in common base fluids of highly enhanced thermal properties. This type of fluids has highly industrial importance because of its unique chemical and physical properties. It has a higher thermal conductivity which controlled significant enhancement due to the rate of heat transfer.

Maxwell fluid model is one of the subclass of rate type fluids which has gained wide attractions among the researchers in last 10 years. The fluid of both viscosity and elasticity properties is so called Maxwell fluid. The main advantage of using this kind of fluids is it can predict the stress relaxation whilst other differential-type fluids cannot predict such effects. Maxwell fluid widely used in the field of viscoelastic fluid in where the relaxation time (dimensionless) is insignificant however, it's beneficial for significant relaxation time in concentrated polymeric fluids of low molecular weight (Fetecau and Zierep, 2003). Hayat and Sajid (2007) studied hydrodynamics of MHD flow of an upper-convected Maxwell fluid by a Homotopy analysis method. The upper-convected Maxwell fluid flow past a porous shrinking sheet was investigated by Hayat et al. (2008). Hayat and coauthors also considers Maxwell fluid flow in a rotating frame (Hayat et al. 2008), over a stretching surface (Hayat et al. 2009), effect of thermal radiation, thermophoresis and Joule heating on Maxwell fluid flow (Havat and Oasim, 2010), channel flow of Maxwell fluid (Havat et al. 2011) in their investigation. Chen et al. (2015) analysed hydrodynamics of Maxwell fluid flow over an unsteady stretching surface. The study of MHD Maxwell fluid flow over a stretching sheet with nanoparticles and presence of chemical reaction was reported by Afify and Elgazery

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(2016). Khan et al. (2016) investigated MHD heat and mass transfer axisymmetric chemically reactive Maxwell fluid flow of driven by exothermal and isothermal stretching disks. In the presence of nano particles, Ramesh et al. (2016) studied Maxwell fluid stagnation point flow of near a permeable surface. Recently, the radiation and viscous dissipation effects on Maxwell fluid flow in a combined MHD heat transfer thermal extrusion system was studied by Hsiao (2017). The rising needs of chemical reactions led the higher importance of investigating chemical reaction influence on MHD heat and mass transfer flow in chemical engineering and hydrometallurgical industries. This study finds vast applications in the field of evaporation, flow in a desert cooler, energy transfer in a cooling tower, drying etc. The effects of chemical reaction on heat and mass transfer MHD boundary layer laminar type flow over a wedge considering suction/injection was analysed by Devi et al. (2002). Kandasamy et al. (2005) investigated the effect of presence of thermal stratification and heat source on chemical reactive fluid flow through a stretching surface. Elgazery (2009) reported the influence of chemical reaction on MHD temperature dependent viscosity and thermal diffusive flow with Hall and ion-slip currents. Zhang and Zheng (2012) examined MHD thermosolutal Marangoni convection flow with incorporation of firstorder chemical reaction. For controlling the hydrodynamics fluid flow behaviour, this study still receiving a lot of consideration in recent years (Eid, 2016; Mabood et al. 2015; Muthuraj et al. 2016; Nayak et al. 2016; Rahman et al. 2014; Reddy, 2016; Reddy et al. 2017; Srinivas et al. 2016).

Now a day, the study of radiation absorption effects on boundary layer fluid flow is industrially important because of manufacturing applications such as catalytic reactors, reactor safety, oil reservoirs, heat insulation, geothermal systems etc. The effect of radiation absorption on transient MHD free convection flow was investigated by Ibrahim et al. (2008). Rubio Hernández (2013) studied the influence of radiation absorption on MHD time dependent fluid flow through a porous medium. MHD micropolar fluid flow in a rotating system with the presence of radiation absorption was investigated by Satya Narayana (2013). Reddy et al. (2015) reported a study concerning with MHD free convective flow with the effect of radiation absorption. More recent studies can have found in the literature (Durga et al. 2016; Satya, 2015; Sreedevi et al. 2016; Umamaheswar et al. 2016).

The effect of heat generation on boundary layer fluid flow is very significant due to engineering applications such as fire and combustion, metal waste, radioactive materials, reactor safety analysis, spent nuclear fuel, etc. Several investigators (Benos et al. 2014; Ganga et al. 2015; Hady et al. 2006; Nandy and Mahapatra, 2013) find out heat generation as an important arena of fluid dynamics research. The heat generation/absorption effects on MHD convective Al₂O₃–water and TiO₂–water nanofluids flow past a stretching sheet in porous media was analysed by Reddy et al. (2016) reported. Srinivasa and Eswara (2016) studied heat generation effect of electrically conducting fluid flow towards an isothermal truncated cone. Hayat et al. (2017) analytically investigated the effect of existence of heat generation/absorption on MHD three-dimensional Oldroyd-B nanofluid flow.

To the best of the author's knowledge, the study of natural convective and chemically reactive Maxwell fluid flow with nanoparticle through a vertical porous plate in presence of thermal radiation, heat generation and radiation absorption has remained unexplored. Therefore, this phenomenon is addressed in this study. The specific objectives of this numerical investigation are listed below:

- a) To investigate unsteady chemically reactive Maxwell fluid flow with nano particle through a vertical porous plate with the influence of thermal radiation, mass diffusion with heat source, mass transfer and radiation absorption.
- b) Mathematical solution of the flow governing model which includes transient momentum, energy and diffusion balance equations numerically using well-known explicit finite difference method (EFDM).

- c) Optimizing the numerical flow parameters and predicting high accuracy of EFDM solutions by analyzing stability and convergence analysis (SCA).
- d) To study velocity, temperature, and concentration distribution across the boundary layer. Investigation on skin friction coefficient, Nusselt number and Sherwood number with different physical parameters.
- e) Evaluation of the thermal and momentum boundary layer thickness with isotherms and streamlines analysis.

2. MATHEMATICAL MODEL OF FLOW

Unsteady heat and mass transfer flow of Maxwell fluid along a semiinfinite vertical porous plate (y = 0) is considered in the presence of a uniform thermal radiation and magnetic field. The fluid is flowing in the x - direction which is taken along the plate in the upward x-direction and y -axis is normal to it. When, the plate velocity U(t) is given as $u = U_0$. In initial step, it is considered that the plate as well as the fluid particle is at rest at the same temperature $T(=T_{\infty})$ and the same concentration level $C(=C_{\infty})$ at all points. Where, C_{∞} and T_{∞} are fluid concentration and temperature species of uniform flow respectively.



Fig. 1. Physical configuration and coordinate system.

It is also assumed that a magnetic field $B_y = B_o$ of uniform strength is applied normal to the flow region. The physical configuration and coordinate system of the problem is presented in the following Fig. 1. Under the assumptions made above, the governing equations for unsteady MHD transient Maxwell fluid flow in presence of Nanoparticle through a vertical porous plate with the effect of chemical reaction, heat generation and radiation absorption into account, reduce to:

$$\frac{\partial \mathbf{u}}{\partial \mathbf{x}} + \frac{\partial \mathbf{v}}{\partial \mathbf{y}} = 0 \tag{1}$$

Momentum Equation,

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = v \frac{\partial^2 u}{\partial x^2} - \lambda_1 \left[u^2 \frac{\partial^2 u}{\partial x^2} + v^2 \frac{\partial^2 u}{\partial y^2} + 2uv \frac{\partial^2 u}{\partial x \partial y} \right]$$

$$- \frac{\sigma B_0^2}{\rho} u + g\beta(T - T_{\infty}) + g\beta^*(C - C_{\infty}) - \frac{v}{k} u$$
(2)

Energy Equation,

$$\frac{\partial T}{\partial t} + u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \frac{\kappa \partial^2 T}{\rho c_p \partial y^2} + \frac{v}{c_p} \left(\frac{\partial u}{\partial y}\right)^2 + \frac{Q_0}{\rho c_p} (T - T_\infty) + \frac{Q_1^*}{\rho c_p} (C - C_\infty) - \frac{1}{\rho c_p} \frac{\partial q_r}{\partial y} + \frac{D_m \kappa_T}{c_s c_p} \frac{\partial^2 C}{\partial y^2} + \frac{\sigma' B_0^2 u^2}{\rho c_p} + \Gamma \left\{ D_B \left(\frac{\partial T}{\partial y} \frac{\partial C}{\partial y}\right) + \frac{D_T}{T_\infty} \left(\frac{\partial T}{\partial y}\right)^2 \right\}$$
(3)

Concentration Equation,

$$\frac{\partial C}{\partial t} + u \frac{\partial C}{\partial x} + v \frac{\partial C}{\partial y} = D_B \left(\frac{\partial^2 C}{\partial y^2} \right) + \frac{D_T}{T_{\infty}} \frac{\partial^2 T}{\partial y^2} - K_c (C - C_{\infty})^P$$
(4)

With boundary condition,

$$\begin{split} &u=U_0, T=T_w, C=C_w \quad \text{at} \quad y=0 \\ &u=0, T=T_\infty, C=C_\infty \quad \text{at} \quad y\to\infty \end{split}$$

Where, u and v are the velocity component, B_0 is the magnetic field component, β is thermal expansion coefficient, β^* is concentration expansion coefficient, T_w denotes the wall temperature, C_w is the species concentration at the wall, v is the kinematic viscosity, ρ is density, κ is thermal conductivity, C_p is specific heat at constant pressure, Q_0 denotes the heat source, Q_1^* denotes the radiation absorption, q_r unidirectional radiative heat flux, K_c for chemical reaction, D_B is Brownian diffusion coefficient, D_T thermophoresis diffusion coefficient and λ_1 is the relaxation time of the upperconvected Maxwell (UCM) fluid. The radiative heat flux term by using the Rosseland approximation is given by $q_r = -\frac{4\sigma_s}{3k_p} \frac{\partial T^4}{\partial y}$.

Where, σ_s is the Stefan-Boltzmann constant and k_e is the mean absorption coefficient, respectively. If temperature differences within the flow are sufficiently small, then the q_r can be linearized by expanding T^4 into the Taylor series about T_{∞} , which after neglecting higher order terms takes the form by $T^4 \cong 4T_{\infty}^3T - 3T_{\infty}^4$. Then the equation (3) becomes,

$$\frac{\partial T}{\partial t} + u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \frac{\kappa \partial^2 T}{\rho c_p \partial y^2} + \frac{v}{c_p} \left(\frac{\partial u}{\partial y}\right)^2 + \frac{Q_0}{\rho c_p} (T - T_{\infty}) + \frac{Q_1^*}{\rho c_p} (C - C_{\infty}) + \frac{16\sigma_s T_{\infty}^3}{3k_e \rho c_p} \frac{\partial^2 T}{\partial y^2} + \frac{D_m \kappa_T}{c_s c_p} \frac{\partial^2 C}{\partial y^2} + \frac{\sigma B_0^2 u^2}{\rho c_p} + \Gamma \left\{ D_B \left(\frac{\partial T}{\partial y} \frac{\partial C}{\partial y}\right) + \frac{D_T}{T_{\infty}} \left(\frac{\partial T}{\partial y}\right)^2 \right\}$$
(5)

From the governing equations (1) - (5) under the initial conditions and the boundary conditions will be based on the finite difference method it is required to make the equations dimensionless. For the purpose introducing the following dimensionless quantities:

$$\begin{aligned} X &= \frac{xU_0}{v}, Y = \frac{yU_0}{v}, U = \frac{u}{U_0}, V = \frac{v}{U_0}, \tau = \frac{tU_0^2}{v}, \theta = \frac{T - T_{\infty}}{T_w - T_{\infty}}, \phi = \frac{C - C_{\infty}}{C_w - C_{\infty}}, \\ \text{So, } x &= \frac{Xv}{U_0}, y = \frac{Yv}{U_0}, u = UU_0, t = \frac{\tau v}{U_0^2}, T = T_{\infty} + \theta(T_w - T_{\infty}) \text{ and} \\ C &= C_{\infty} + \phi(C_w - C_{\infty}) \end{aligned}$$

the dimensionless equations are obtained as follows:

Dimensionless Continuity Equation,

$$U\frac{\partial U}{\partial X} + V\frac{\partial U}{\partial Y} = 0 \tag{6}$$

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Dimensionless Momentum Equation,

$$\frac{\partial U}{\partial \tau} + U \frac{\partial U}{\partial X} + V \frac{\partial U}{\partial Y} = \frac{\partial^2 U}{\partial Y^2} - N_v \left[U^2 \frac{\partial^2 U}{\partial X^2} + V^2 \frac{\partial^2 U}{\partial Y^2} + 2UV \frac{\partial^2 U}{\partial X \partial Y} \right]_{(7)}$$
$$+ G_r \theta + G_c \phi - MU - \frac{1}{D_a} U$$

Dimensionless Energy Equation,

$$\frac{\partial\theta}{\partial\tau} + \frac{\partial\theta}{\partial Y}U + V\frac{\partial\theta}{\partial Y} = \frac{1}{P_r} \left(1 + \frac{16R}{3}\right) \frac{\partial^2\theta}{\partial Y^2} + Q\theta + Q_l\phi + E_c \left(\frac{\partial U}{\partial Y}\right)^2 + D_u \frac{\partial^2\phi}{\partial Y^2} \quad (8)$$
$$+ E_c M U^2 + N_b \left(\frac{\partial\theta}{\partial Y}\frac{\partial C}{\partial Y}\right) + N_t \left(\frac{\partial\theta}{\partial Y}\right)^2$$

Dimensionless Concentration Equation,

$$\frac{\partial \phi}{\partial \tau} + U \frac{\partial \phi}{\partial X} + V \frac{\partial \phi}{\partial Y} = \frac{1}{L_e} \left[\frac{\partial^2 \phi}{\partial Y^2} + \left(\frac{N_t}{N_b} \right) \frac{\partial^2 \theta}{\partial Y^2} \right] - K_r \phi^P$$
(9)

$$U = 1, \ \theta = 1, \ \varphi = 1 \quad \text{at} \quad y = 0$$

$$U = 0, \ \theta = 0, \ \varphi = 0 \quad \text{at} \quad y \to \infty$$

Where the obtained physical parameters are given below:

magnetic parameter,
$$M = \frac{\sigma' B_0^2 v}{\rho U_0^2}$$
, Darcy number, $D_a = \frac{K' U_0^2}{v^2}$, Prandtl

number, $P_r = \frac{\rho c_p \upsilon}{\kappa}$, radiation parameter, $R = \frac{\sigma T_{\infty}^3}{k_e \kappa}$, heat source

parameter,
$$Q = \frac{Q_0 v}{U_0^2 \rho c_p}$$
, radiation absorption parameter, $Q_1 = \frac{Q_1^* v}{U_0^2 \rho c_p}$,

Eckert number,

number, $E_c = \frac{U_0^2}{c_p(T_w - T_\infty)}$, Dufour number,

$$D_u = \frac{D_m \kappa_T}{c_s c_p v} \left(\frac{C_w - C_\infty}{T_w - T_\infty} \right), \quad \text{Lewis number,} \quad L_e = \frac{v}{D_m}, \text{Maxwell}$$

parameter, $N_v = \frac{\lambda_l U_0^2}{v}$, Brownian parameter, $N_b = \frac{\Gamma D_B (C_w - C_\infty)}{v}$,

thermophoresis parameter, $N_t = \frac{\Gamma D_T}{T_{\infty} \upsilon} (T_w - T_{\infty})$, Order of chemical

reaction and = P, chemical reaction, $K_r = \frac{\nu K_c (C_w - C_\infty)^{p-1}}{U_0^2}$.

Stream function Ψ satisfies the continuity Eq. (6) and is associated with the velocity components in the usual way as,

$$U = \frac{\partial \psi}{\partial Y}, V = -\frac{\partial \psi}{\partial X}$$

The parameters of technological interest for the present problem are the local skin-friction, the local Nusselt number and the local Sherwood number, which are elucidated as;

$$C_f = -\frac{1}{2\sqrt{2}} G_r^{-3/4} \left(\frac{\partial U}{\partial Y}\right)_{Y=0}$$
(10)

$$N_u = \frac{1}{\sqrt{2}} G_r^{-3/4} \left(\frac{\partial \theta}{\partial Y}\right)_{Y=0}$$
(11)

$$S_h = \frac{1}{2\sqrt{2}} G_r^{-3/4} \left(\frac{\partial\phi}{\partial Y}\right)_{Y=0}$$
(12)

3. NUMERICAL TECHNIQUE

To solve the governing coupled non-dimensional partial differential equations with the associated initial and boundary conditions. The method of explicit finite difference has been used to solve (6) - (9) subject to the initial and boundary conditions. For this reason, the area

within the boundary layer is divided by some perpendicular lines of Y - axis, where the normal of the medium is Y - axis as shown in Fig. 2. It is assumed that the maximum length of boundary layer $Y_{\text{max}} = 20$ as corresponds to $Y \rightarrow \infty$. i.e. Y vary from 0 to 20 and the number of grid spacing in Y directions are m(=100) and n(=200), with the smaller time step $\Delta \tau = 0.005$. Using the explicit finite difference approximation, we have,

Continuity Equation,

$$\frac{U_{i,j} - U_{i-1,j}}{\Delta X} + \frac{V_{i,j} - V_{i,j-1}}{\Delta Y} = 0$$
(13)

Momentum Equation

$$\frac{U_{i,j}^{'} - U_{i,j}}{\Delta \tau} + U_{i,j} \frac{U_{i,j} - U_{i-1,j}}{\Delta X} + V_{i,j} \frac{U_{i,j+1} - U_{i,j}}{\Delta Y} = \frac{U_{i,j+1} - 2U_{i,j} + U_{i,j-1}}{(\Delta Y)^{2}} \\
-N_{\nu} [(U_{i,j})^{2} \frac{U_{i,j+1} - 2U_{i,j} + U_{i,j-1}}{(\Delta X)^{2}} + (V_{i,j})^{2} \frac{U_{i,j+1} - 2U_{i,j} + U_{i,j-1}}{(\Delta Y)^{2}} \\
+ (2U_{i,j}V_{i,j} \frac{U_{i+1,j+1} - U_{i+1,j-1} - U_{i-1,j+1} + U_{i-1,j-1}}{4\Delta X \Delta Y})] \\
+ G_{r} \theta_{i,j} + G_{c} \phi_{i,j} - (M + \frac{1}{D_{a}})U_{i,j}$$
(14)

Energy Equation,

$$\frac{\theta_{i,j} - \theta_{i,j}}{\Delta \tau} + U_{i,j} \frac{\theta_{i,j} - \theta_{i-1,j}}{\Delta X} + V_{i,j} \frac{\theta_{i,j+1} - \theta_{i,j}}{\Delta Y} = \frac{1}{P_r} (1 + \frac{16}{3} R) \frac{\theta_{i,j+1} - 2\theta_{i,j} + \theta_{i,j-1}}{(\Delta Y)^2}$$

$$+ D_u \frac{\phi_{i,j+1} - 2\phi_{i,j} + \phi_{i,j-1}}{(\Delta Y)^2} + Q\theta_{i,j} + Q_l \phi_{i,j} + E_c \left(\frac{U_{i,j+1} - U_{i,j}}{\Delta Y}\right)^2 + E_c M \left(U_{i,j}\right)^2$$

$$+ N_b \left(\frac{\theta_{i,j+1} - \theta_{i,j}}{\Delta Y} \cdot \frac{\varphi_{i,j+1} - \varphi_{i,j}}{\Delta Y}\right) + N_r \left(\frac{\theta_{i,j+1} - \theta_{i,j}}{\Delta Y}\right)^2$$

$$(15)$$

Concentration Equation,

$$\frac{\phi_{i,j} - \phi_{i,j}}{\Delta \tau} + U_{i,j} \frac{\phi_{i,j} - \phi_{i-l,j}}{\Delta X} + V_{i,j} \frac{\phi_{i,j+1} - \phi_{i,j}}{\Delta Y} = \frac{1}{L_e} \left[\frac{\phi_{i,j+1} - 2\phi_{i,j} + \phi_{i,j-1}}{(\Delta Y)^2} \right] + \left(\frac{N_t}{N_b} \frac{\theta_{i,j+1} - 2\theta_{i,j} + \theta_{i,j-1}}{(\Delta Y)^2} \right] - K_r(\phi_{i,j})^P$$
(16)



Fig. 2 The finite difference space grid.

The initial and boundary condition with finite difference scheme as $U_{i,0}^n = 1, \theta_{i,0}^n = 1, \phi_{i,0}^n = 1$

 $U_{i,L}^{n} = 0, \ \theta_{i,L}^{n} = 0, \ \phi_{i,L}^{n} = 0$ Where, $L \to \infty$

Where, the subscripts *i* and *j* designate the grid points with *X* and *Y* coordinates respectively and a value of time, $\tau = n\Delta\tau$, where $n = 1, 2, 3, 4, \dots$.

4. STABILITY AND CONVERGENCE ANALYSIS OF THE PROBLEM

For the solution an explicit finite difference approach is being used therefore the analysis is remained incomplete unless the stability and convergence of the finite difference scheme are discussed. For the constant mesh size, the stability criteria of the scheme may be established as follows. The general terms of the Fourier expansion for U, θ and φ at a time arbitrarily called t = 0 are all $e^{i\alpha x} e^{i\beta y}$ apart from a constant, where $i = \sqrt{-1}$. A time $t = \tau$, these terms becomes

$$\left. \begin{array}{c} U: \psi(\tau) e^{i\alpha X} e^{i\beta Y} \\ g: \theta(\tau) e^{i\alpha X} e^{i\beta Y} \\ p: \varphi(\tau) e^{i\alpha X} e^{i\beta Y} \end{array} \right\}$$

$$(17)$$

after a time step these terms convert to

$$\begin{array}{c} U: \psi'(\tau) e^{i\alpha x} e^{i\beta Y} \\ \theta: \theta'(\tau) e^{i\alpha x} e^{i\beta Y} \\ \varphi: \varphi(\tau) e^{i\alpha x} e^{i\beta Y} \end{array}$$

$$(18)$$

Substituting (17) and (18) to the main (13) -(16) equation we get,

$$\begin{split} \psi' &= \psi(\tau) + \Delta\tau [G_r \theta + G_c \varphi - (M + \frac{1}{D_a})\psi + \frac{2(\cos\beta\Delta Y - 1)}{(\Delta Y)^2}\psi \\ &- \frac{U(1 - e^{i\alpha\Delta X})}{\Delta X}\psi - \frac{V(e^{i\beta\Delta Y} - 1)}{\Delta Y}\psi - N_v [\frac{2U^2\psi(1 - \cos\alpha\Delta X)}{(\Delta X)^2} \\ &+ \frac{2V^2\psi(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} \\ &+ 2\psi UV \frac{e^{i\alpha(X + \Delta X)}e^{i\beta(Y + \Delta Y)} - e^{i\alpha(X + \Delta X)}e^{i\beta(Y - \Delta Y)}}{4\Delta X \Delta Y} \\ &- 2\psi UV \frac{(e^{i\alpha(X - \Delta X)}e^{i\beta(Y + \Delta Y)} - e^{i\alpha(X - \Delta X)}e^{i\beta(Y - \Delta Y)})}{4\Delta X \Delta Y}]]] \\ \Rightarrow \psi' = [1 + \Delta\tau \frac{2(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} - \Delta\tau (M + \frac{1}{D_a}) - \frac{U\Delta\tau (1 - e^{i\alpha\Delta X})}{\Delta X} \\ &- \frac{V\Delta\tau (e^{i\beta\Delta Y} - 1)}{\Delta Y} + \Delta\tau G_r \theta + \Delta\tau G_c \varphi \\ &- N_v [\frac{2U^2\Delta\tau (1 - \cos\alpha\Delta X)}{(\Delta X)^2} + \frac{2V^2\Delta\tau (\cos\beta\Delta Y - 1)}{(\Delta Y)^2} \\ &+ 2UV\Delta\tau \frac{e^{i\alpha(X + \Delta X)}e^{i\beta(Y + \Delta Y)} - e^{i\alpha(X + \Delta X)}e^{i\beta(Y - \Delta Y)}}{4\Delta X \Delta Y}]]]\psi \end{split}$$

$$\psi' = A_1 \psi + A_2 \theta + A_3 \varphi$$

(19)

where

$$\begin{split} A_{1} &= 1 + \Delta \tau \frac{2(\cos\beta\Delta Y - 1)}{(\Delta Y)^{2}} - \Delta \tau (M + \frac{1}{D_{a}}) - \frac{U\Delta \tau (1 - e^{i\alpha\Delta X})}{\Delta X} \\ &- \frac{V\Delta \tau (e^{i\beta\Delta Y} - 1)}{\Delta Y} \\ &- N_{v} [\frac{2U^{2}\Delta \tau (1 - \cos\alpha\Delta X)}{(\Delta X)^{2}} + \frac{2V^{2}\Delta \tau (\cos\beta\Delta Y - 1)}{(\Delta Y)^{2}} \\ &+ 2UV\Delta \tau \frac{e^{i\alpha(X + \Delta X)}e^{i\beta(Y + \Delta Y)} - e^{i\alpha(X + \Delta X)}e^{i\beta(Y - \Delta Y)}}{4\Delta X\Delta Y} \\ &- 2UV\Delta \tau \frac{(e^{i\alpha(X - \Delta X)}e^{i\beta(Y + \Delta Y)} - e^{i\alpha(X - \Delta X)}e^{i\beta(Y - \Delta Y)})}{4\Delta X\Delta Y}] \\ , A_{2} &= \Delta \tau G_{r} \text{ and } A_{3} = \Delta \tau G_{c} \end{split}$$

For temperature equation,

$$\begin{split} \theta' &= \theta + \Delta \tau \Big[\frac{1}{P_r} (1 + \frac{16R}{3}) \frac{2\theta(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} + D_u \frac{2\varphi(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} + Q_1 \varphi \\ &+ E_c \frac{\left(e^{i\beta\Delta Y} - 1\right)^2}{\Delta Y^2} \psi + Q\theta - \frac{U(1 - e^{i\alpha\Delta X})}{\Delta X} \psi - \frac{V(e^{i\beta\Delta Y} - 1)}{\Delta Y} \psi \\ &+ E_c M \psi^2 + N_b \theta C \left\{ \frac{e^{i\beta\Delta Y} - 1}{\Delta Y} \right\}^2 + N_t T \theta \left\{ \frac{e^{i\beta\Delta Y} - 1}{\Delta Y} \right\}^2 \Big] \\ \theta' &= \theta \Big[1 + \frac{\Delta \tau}{P_r} (1 + \frac{16R}{3}) \frac{2(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} + Q\Delta \tau - \frac{U\Delta \tau (1 - e^{i\alpha\Delta X})}{\Delta X} \\ &- \frac{V\Delta \tau (e^{i\beta\Delta Y} - 1)}{\Delta Y} + N_t T \Delta \tau \left\{ \frac{e^{i\beta\Delta Y} - 1}{\Delta Y} \right\}^2 + N_b C \Delta \tau \left\{ \frac{e^{i\beta\Delta Y} - 1}{\Delta Y} \right\}^2 \Big] \\ &+ \left[Q_1 \Delta \tau + D_u \frac{2\Delta \tau (\cos\beta\Delta Y - 1)}{(\Delta Y)^2} \right] \varphi + E_c \Delta \tau \frac{U(e^{i\beta\Delta Y} - 1)^2}{(\Delta Y)^2} \psi + E_c \Delta \tau M \psi^2 \end{split}$$

For obtaining the stability condition, it should be find out Eigen values of the amplification matrix T, but this study is very difficult since all the elements of T are different. Hence, the problem requires that the Eckert number E_c is assumed to be very small, that is, tends to zero. Then we get,

$$\theta' = \theta \left[1 + \frac{\Delta \tau}{P_r} \left(1 + \frac{16R}{3}\right) \frac{2(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} + Q\Delta \tau - \frac{U\Delta \tau (1 - e^{i\alpha\Delta X})}{\Delta X} - \frac{V\Delta \tau (e^{i\beta\Delta Y} - 1)}{\Delta Y} + N_r T\Delta \tau \left\{\frac{e^{i\beta\Delta Y} - 1}{\Delta Y}\right\}^2 + N_b C\Delta \tau \left\{\frac{e^{i\beta\Delta Y} - 1}{\Delta Y}\right\}^2\right] + \left[Q_1\Delta \tau + D_u \frac{2\Delta \tau (\cos\beta\Delta Y - 1)}{(\Delta Y)^2}\right]\varphi$$

$$\theta' = A_4\theta + A_5\varphi \tag{20}$$

Where,

$$A_{4} = 1 + \frac{\Delta\tau}{P_{r}} (1 + \frac{16R}{3}) \frac{2(\cos\beta\Delta Y - 1)}{(\Delta Y)^{2}} + Q\Delta\tau - \frac{U\Delta\tau(1 - e^{i\alpha\Delta X})}{\Delta X} - \frac{V\Delta\tau(e^{i\beta\Delta Y} - 1)}{\Delta Y} + N_{t}T\Delta\tau \left\{\frac{e^{i\beta\Delta Y} - 1}{\Delta Y}\right\}^{2} + +N_{b}C\Delta\tau \left\{\frac{e^{i\beta\Delta Y} - 1}{\Delta Y}\right\}^{2}$$

and $A_{5} = Q_{1}\Delta\tau + D_{u}\frac{2\Delta\tau(\cos\beta\Delta Y - 1)}{(\Delta Y)^{2}}$.

For the concentration equation,

$$\varphi' = \varphi + \Delta \tau \left[\frac{1}{L_e} \left[\frac{2\varphi(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} + \frac{N_t}{N_b} \frac{2(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} \theta \right] - K_r \varphi^p - \frac{U(1 - e^{i\alpha\Delta X})}{\Delta X} \varphi - \frac{V(e^{i\beta\Delta Y} - 1)}{\Delta Y} \varphi \right]$$

(21)

$$\varphi' = \varphi \left[1 + \frac{2\Delta\tau}{S_c} \frac{(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} - \Delta\tau K_r - \frac{U\Delta\tau(1 - e^{i\alpha\Delta X})}{\Delta X} - \frac{V\Delta\tau(e^{i\beta\Delta Y} - 1)}{\Delta Y}\right] + \frac{N_t}{N_b} \frac{2(\cos\beta\Delta Y - 1)}{(\Delta Y)^2} \theta$$

 $\varphi' = A_6 \varphi + A_7 \theta$ Where.

$$A_{6} = 1 + \frac{2\Delta\tau}{L_{e}} \frac{(\cos\beta\Delta Y - 1)}{(\Delta Y)^{2}} - \Delta\tau K_{r} - \frac{U\Delta\tau(1 - e^{i\alpha\Delta X})}{\Delta X} - \frac{V\Delta\tau(e^{i\beta\Delta Y} - 1)}{\Delta Y} \text{ and}$$
$$A_{7} = \left(\frac{N_{t}}{N_{b}}\right) \frac{2(\cos\beta\Delta Y - 1)}{(\Delta Y)^{2}}.$$

Equation (19) -(21) can be expressed in the Matrix form,

$$\begin{bmatrix} \boldsymbol{\psi}'\\ \boldsymbol{\theta}'\\ \boldsymbol{\varphi}'\end{bmatrix} = \begin{bmatrix} A_1 & A_2 & A_3\\ 0 & A_4 & A_5\\ 0 & A_7 & A_6 \end{bmatrix} \begin{bmatrix} \boldsymbol{\psi}\\ \boldsymbol{\theta}\\ \boldsymbol{\varphi} \end{bmatrix} \text{ i.e. } \boldsymbol{\eta}' = T'\boldsymbol{\eta}$$

For obtaining the stability condition, Eigen values of the amplification matrix T' should be finding out. It is a forth order square matrix. For this explicit finite difference solution, the dimensionless time difference $\Delta \tau$ is very small i.e. tends to zero. Under this condition,

$$A_2 \to 0, A_3 \to 0, A_5 \to 0 \text{ and } A_7 \to 0$$

 $\therefore T' = \begin{bmatrix} A_1 & 0 & 0 \\ 0 & A_4 & 0 \\ 0 & 0 & A_6 \end{bmatrix}$

After simplification of the matrix T' the Eigen values are follows, The Eigen values of the amplification matrix T^* are obtained as $A_1 = \lambda_1, A_4 = \lambda_2$ and $A_6 = \lambda_3$. For stability test, each of the Eigen values must not exceeded unity in modulus. Under this consideration, the stability conditions are as follows

$$|A_1| \le 1, |A_4| \le 1$$
 and $|A_6| \le 1$
Let, $a_1 = \Delta \tau, b_1 = U \frac{\Delta \tau}{\Delta X}, c_1 = |-V| \frac{\Delta \tau}{\Delta X}$ and $d_1 = 2 \frac{\Delta \tau}{(\Delta Y)^2}$ then,

The coefficient of a, b and c are all non-negative. So the maximum modulus of A_1 , A_4 and A_6 occurs when $\alpha \Delta Y = n\pi$, where *n* is integer and hence A_1 , A_4 and A_6 are real. The values of $|A_1|$, $|A_4|$ and $|A_6|$ are greater when *n* is odd integer, in which case;

$$A_{4} = 1 - 2 \left[d_{1} \frac{1}{P_{r}} \left(1 + \frac{4}{3}R \right) + \frac{a_{1}}{2}Q + b_{1} + c_{1} + 2N_{b}Cd_{1} + 2N_{r}Td_{1} \right]$$
$$A_{6} = 1 - 2 \left[d_{1} \frac{1}{L_{e}} + b_{1} + c_{1} + \frac{a_{1}}{2}K_{c} \right]$$

To satisfied allowable values are $A_1 = -1$ $A_4 = -1$ and $A_6 = -1$. Hence the stability conditions of the methods are,

$$U\frac{\Delta\tau}{\Delta X} + V\frac{\Delta\tau}{\Delta X} + \frac{2}{P_r}\left(1 + \frac{16}{3}R\right)\frac{\Delta\tau}{\left(\Delta Y\right)^2} - \frac{\Delta\tau Q}{2} + 2N_b\frac{\Delta\tau}{\left(\Delta Y\right)^2} + 2N_r\frac{\Delta\tau}{\left(\Delta Y\right)^2} \le 1$$

and, $U\frac{\Delta\tau}{\Delta X} + V\frac{\Delta\tau}{\Delta X} + \frac{2}{L_e}\frac{\Delta\tau}{\left(\Delta Y\right)^2} + \frac{\Delta\tau K_r}{2} \le 1$.

With initial boundary conditions U = V = T = C = 0 and for the values of $\Delta \tau = 0.005$, $\Delta X = 0.20$ and $\Delta Y = 0.25$ then the problem will be converged at $P_r \ge 0.209$ and $L_e \ge 0.16$. These converge solutions are shown graphically in Figs. 3-19.

5. RESULTS AND DISCUSSION

To investigate the physical phenomena of the problem, a finite difference solution is obtained by the use of an explicit procedure. The

numerical values of non-dimensional velocity, temperature and concentration within the boundary layer for different values of nondimensional parameter have been computed by a FORTRAN program. For the steady-state solutions, the computations have been carried out up to dimensionless time $\tau = 30$. (see Appendix A) Porous flow is always very with low Reynolds number so any inertial effects are practically non-existent. Porosity to gases are different than those for liquids in the same media. In Fig. 3(a) and Fig. 3(b)the velocity and skin friction profiles increase in case of strong Darcy number (D_a) and become zero and for the skin friction profiles it remains unchanged with in short range for same values of Darcy number (D_a). This parameter has a very strong physical significance on the fluid velocity. For the values of $\mathrm{D_a} \geq 0.50$, the fluid velocity increases rapidly. The numerical value obtained from the EFDM simulation for Darcy number (D_a) varies from 0.50 to 3.0 the curve to curve fluctuations of velocity profiles are 207.328 %, 76.833% and 28.597% increase at point Y = 3.00. On the other hand, skin friction increases with 63.52 %, 26.562% and 11.737% at $\tau = 10$.



Fig. 3 Illustration of (a) velocity profiles for different values of D_a and (b) skin friction for different values of D_a .

In Fig. 4(a) and Fig. 4(b) illustrate the variations of concentration and temperature profiles for various values of Brownian parameter (N_b) . It is found that the increase in the value of $N_b \ge 0.10$, the concentration boundary layer reduces and the thermal boundary layer increases. The numerical value obtained from the EFDM simulation for Brownian parameter (N_b) varies from 0.10 to 0.60. The curve to curve increment at Y = 2.00 are 50.22 %, 13.81% and 4.82% increase for concentric boundary layer. On the other hand, thermal layer fluctuates 120.505 %, 87.965% and 49.74% increase for Brownian parameter (N_b) varies from 0.10 to 0.60.



Fig. 4 Illustration of (a) concentration profiles for different values of N_b and (b) temperature for different values of N_b .

In Fig. 5(a) and Fig. 5(b) illustrate the variations temperature and concentration profiles. With the increase of thermophoresis parameter $N_t \ge 0.10$, the thermal boundary layer profiles decrease. On the other hand, concentration boundary layer profiles increase rapidly. The first two curve fluctuation for the thermophoresis parameter (N_t) variation is more than 100%.



Fig. 5 Illustration of (a) temperature profiles for different N_t and (b) concentration profiles for different values of N_t .

Table 1 provides the numerical values of the local Nusselt number (Eq. (11)) obtained from the EFDM simulation for different thermophoresis parameter (N_t) by changing N_b . The local Nusselt number is observed to be increase with increasing values of N_t and N_b .

Table 1. Computations are showing the increase of Nusselt number (N_u) for the increase of N_t for $P_r = 1.38$, $L_e = 1.00$ and $\tau = 1.20$.

N_t	N_u	N_{u}	N_{u}	N_u	N_{u}
	$(N_{b} =$				
	0.10)	0.20)	0.30)	0.40)	0.50)
0.10	0.07855	0.00310	-0.02716	-0.02451	-0.02100
0.20	0.14424	0.05009	-0.01396	-0.03049	-0.01976
0.30	0.21444	0.11994	0.03272	-0.01688	-0.01449
0.40	0.28348	0.19285	0.09963	0.01639	-0.01070
0.50	0.33313	0.25967	0.17173	0.08377	0.00461

Table 2 provides the numerical values of local Sherwood number (Eq. (12)) obtained from the EFDM simulation for different thermophoresis parameter (N_t) by changing N_b . The local Sherwood number (S_h) is decreased with the increment of N_t and N_b .

Table 2. Computations are showing the decreased of Sherwood number (S_k) for the increase of N_r for $P_r = 1.38$, $L_r = 1.00$ and $\tau = 1.20$.

		1 1			
N_t	S_h	S_h	S_h	S_h	S_h
	$(N_{b} =$				
	0.10)	0.20)	0.30)	0.40)	0.50)
0.10	0.17240	0.20418	0.29791	0.29378	0.28957
0.20	0.06532	0.19508	0.22023	0.21174	0.24131
0.30	-0.15078	0.12756	0.21808	0.20774	0.22378
0.40	-0.46245	-0.00146	0.16036	0.20217	0.21508
0.50	-0.81565	-0.17724	0.06218	0.18273	0.20686



Fig. 6 Illustration of (a) velocity profiles for different values of N_{ν} and (b) skin friction profiles for different values of N_{ν} .

In Fig. 6(a) and Fig. 6(b) illustrate the variations velocity and skin friction profiles with the effect of Maxwell parameter (N_v). It can be seen that, N_v has a significant impact on velocity boundary layer change. With the increase of N_v , the velocity and skin friction profiles increase rapidly. It is about more than 3000% for velocity and 1000% for skin friction profiles for the curve to curve fluctuation. This behavior occurred because of the Skin friction or friction loss is the loss of pressure that happens in pipe or duct flow for the effect of the fluid's viscosity near the surface of the pipe or duct. In the skin friction, the profiles pick up below to X = 1.2 point and then remain unchanged.



Fig. 7 Illustration of (a) velocity profiles for different values of *M* and (b) skin friction profiles for different values of *M*.

The interaction of electrically conducting fluids with magnetic fields, through electromagnetic forces called Lorentz forces. Strong magnetic parameter ($M \ge 0.30$) creates drag force known as Lorentz force and the force impact the fluid velocity to decrease.

In Fig. 7(a) and Fig. 7(b) exhibits the velocity and skin friction profiles for different M. For the strong magnetic parameter, the velocity and skin friction profiles decrease because of the drag force effect. An immense change of M creates vast fluctuation of curve. For changing magnetic parameter 54.48 %, 38.223% and 29.72% decrease for curve to curve at Y= 3.00. Also for the skin friction, the curves fluctuate occurs 22.85 %, 15.156% and 11.866% decrease at $\tau = 10.00$.

In Fig. 8(a) and Fig. 8(b), the variations of concentration and Sherwood number profiles are shown for different values of K_r . For the increase of destructive chemical reaction ($K_r \ge 0.50$) parameter, the concentration boundary layer profiles decrease, and Sherwood number profiles increase. Increase of K_r significantly alerts the concentration boundary layer thickness but does not alerts momentum boundary layer.

Fluid has two different transport phenomena, one is transport due to convection for the macroscopic fluid velocity and other is transport due to diffusion. In fluids both transport phenomena occur simultaneously. Viscosity is the main element for the fluid transport

phenomenon. Lewis number (L_e) is the ratio of the thermal diffusivity to mass diffusivity.



Fig. 8 Illustration of (a) concentration profiles for different K_r and (b) Sherwood number profiles for different K_r .



Fig. 9 Illustration of (a) concentration profiles for different L_e and (b) Sherwood number profiles for different L_e .

Fig. 9(a) and Fig. 9(b) illustrates the variations of concentration and local Sherwood number profiles for different values of Lewis number, L_e . For the increase of L_e , concentration boundary layer

profiles observed to be decrease rapidly whilst the local Sherwood number profiles increase. The numerical value obtained from the EFDM simulation for Lewis number in the curve to curve fluctuations decrease with 24.34 %, 11.704% and 5.654% at Y= 3.00.



Fig. 10 Illustration of (a) concentration profiles for different values of P and (b) Nusselt number profiles for different D_u .

In Fig. 10(a) and Fig. 10(b) provides the variations of concentration and Nusselt number profiles for different values of P and D_{μ} respectively. Order of the destructive chemical reaction (P) is observed to be influence the concentration profiles as it is increase due to increase in P. On the other hand, Mass diffusion gradient has a strong impact on heat transport phenomena. With the increase of D_{μ} , Nusselt number profiles is found to be decrease. At Y =3.00, the concentric boundary layer fluctuations curve to curve are 6.406 %, 3.069% and 1.614% increase for order of the destructive chemical reaction. On the other hand, Nusselt number (N_u) decrease around 3.09%, 36.63% and 163.71% for the increase of mass diffusion parameter. Fig. 11(a) - Fig. 12(a) displays the variations of velocity, temperature and skin friction profiles for different values of thermal radiation (R) parameter. With the increase of R, the velocity, temperature and skin friction profiles are also increase. Thermal radiation (electromagnetic radiation) could be attributed due to thermal excitation. The temperature could be affected in presence of thermal radiation at moderate temperatures which is significant. Thermal radiation for a medium which contains it inevitably has pressure and density gradients and the treatment requires the use of hydrodynamics. For the electromagnetic radiation by the thermal excitation increase the fluid velocity and it is maximum to near the plate. For the increment of R, the velocity curves fluctuate 279.94 %, 243.34 % and 209.43% and increase at a point Y=3.00. On the other hand, thermal boundary layer fluctuates 51.28 %, 44.65% and 37.82%.

Fig. 12(b) - Fig. 13(b) represent the effect of Prandtl number (P_r) on temperature, local Nusselt and Sherwood number distributions. The parameter, P_r is the ratio of kinematics viscosity to the thermal diffusivity which is physically very with temperature for example, water $P_r = 7.0$ (At 20 °C), Ammonia gases $P_r = 1.38$ falls more quickly compared to air $P_r = 0.71$. In addition, $P_r \ll 1$ explains the thermal

diffusivity dominates. For the large values of Prandtl number, i.e., $P_r >> 1$, the momentum diffusivity dominates this behavior. With the increase of Prandtl number, temperature and Nusselt number (N_u) profiles decrease whereas Sherwood number shows reverse properties.



Fig. 11 Illustration of (a) velocity profiles for different values of *R* and (b). temperature profiles for different values of *R*.



Fig. 12 Illustration of (a) skin friction profiles for different values of R and (b) temperature profiles for different values of P_r .



Fig. 13 Illustration of (a) Nusselt number profiles for different P_r and (b) Sherwood number profiles for different P_r .



Fig. 14 Illustration of (a) Nusselt number profiles for different Q and (b) Nusselt number profiles for different Q₁.

Fig. 14(a) - Fig. 14(b) represent the effect of heat generation (Q) and radiation absorption (Q_1) on Nusselt number profiles. With the increase of heat source and radiation absorption, Nusselt number profiles are observed to be decrease. With the numerical value obtained from the EFDM simulation it was observed that at $\tau = 10.00$ dimensionless time, Nusselt number curves decrease at the rate of 110.8 %, 43.14 % and 43.30% for increasing Q whereas it decreases at the rate of 29.38 %, 25.12% and 20.52% for Q_1 .



Fig. 15 Illustration of (a) Isotherms for R = 0.05 (red solid line) and R = 0.50 (green dashed dot line) and (b) Isotherms flood view.

The non-dimensional equation after different transformations has been solved in the present numerical study. Therefore, for this reason, X and Y axis are dimensionless which indicates the mash point different from the numerical point of view. In addition, with the stream and isotherms (line view) curves, the difference of boundary layer for different parameters can be defined. The development of streamlines and isotherms are presented in Figs. 15-18. It can be observed that; thermal boundary layer and momentum boundary layer increases due to the increase of thermal radiation (R). The legend values of stream and isotherms (flood view) indicate the contours levels. Furthermore, streamlines profiles can be used to improved visualization of fluid fields. It represents the velocity direction of fluids correspondingly. The streamlines can be obtained by drawing lines tangent to the flags. They are a rather mathematical object. They can be visualized by implanting little flags inside the fluid and observing their orientation. The boundary layer system of change can be shown by an isotherm, where the temperature remains constant ($\Delta T = 0$). An isotherm at $0^{\circ}C$ (the freezing point of water) is called the freezing level.

This phenomenon occurs when a system is in contact with an outside thermal layer (heat bath), and the change happens gradually enough to allow the system to continually adjust to the temperature of the boundary layer through heat exchange. Fig. 15 (a) displays the thermal direction of fluid (thermal boundary layer thickness) with the effect of presence of two different thermal radiation parameters. Where the red line indicates R = 0.05 and green line is for R = 0.50. Fig. 15 (b) shows the contour flood view for thermal radiation difference. The contours levels are also provided in the legend of Fig. 15 (b).

Fig. 16 (a) exhibits the isotherms plot of the momentum boundary layer thickness with the effect of presence of thermal radiation parameter where the red line is for R= 0.05 and green line is for R= 0.50. Fig. 16 (b) shows the contour flood view for thermal radiation difference. The legend values of Fig. 16 (b) are given in the contours levels. The thermal and momentum boundary layer thickness is observed to be increases due to increase in R.



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Fig. 16 Illustration of (a) Streamlines for R = 0.05 (red solid line) and R = 0.50 (green dashed dot line) and (b) Streamlines flood view.



Fig. 17 Illustration of (a) Isotherms for $P_r = 1.38$ (red solid line) and $P_r = 7.00$ (green dashed dot line) and (b) Isotherms flood view.

Fig. 17 (a) shows the thermal direction of fluid (thermal boundary layer thickness) with different Prandtl number (P_r). The red line indicates $P_r = 1.38$ whereas green line is for $P_r = 7.00$. Fig. 17 (b) shows the contour flood view of Fig. 17 (a). The legend values of Figs. 17 (b) are the contours levels.



Fig. 18 Illustration of (a) Streamlines for P_r = 1.38 (red solid line) and P_r = 7.00 (green dashed dot line) and (b) Streamlines flood view.

Fig. 18 (a) shows the velocity direction of fluid (momentum boundary layer thickness) with the Prandtl number (P_r) difference. In the graph, red line is for $P_r = 1.38$ and green line is for $P_r = 7.00$. Fig. 18 (b) shows the contour flood view for Prandtl number (P_r) difference. The legend values of Fig. 18 (b) are also given in the contours levels. For the change of Prandtl number (P_r) the thermal and momentum boundary layer thickness decrease.

6. CONCLUDING REMARKS

The Numerical solutions for Maxwell fluid with nano-particle towards a moving semi-infinite porous plate with thermal radiation, heat source, chemical reaction, mass diffusion is analyzed. The results are presented graphically with various parameters. Form the graphical representation, we have the following observations:

- Velocity profiles increase for increasing Darcy number, Maxwell parameter and thermal radiation individually however it decreases due to increase in magnetic parameter and Prandtl number.
- Temperature profiles found increase for individual rising of Brownian parameter, thermal radiation whilst it decreased due to increase in thermophoresis parameter and Prandtl number.
- Concentration profiles increase with the increase of individual Brownian parameter, thermophoresis parameter and chemical reaction order whereas profiles decrease with the rise of chemical reaction, Lewis number separately.
- Skin friction profiles increase for increasing of Darcy number, Maxwell parameter and thermal radiation respectively whilst it decreases with the increase magnetic parameter.
- Sherwood number profiles observe to be increased for the individual rising effects of chemical reaction, Lewis number and Prandtl number.
- Nusselt number profiles decrease for increasing mass diffusion, Prandtl number and heat and radiation absorption individually.

NOMENCLATURE

B_{\circ} C_{f}	magnetic component, (Wb m ⁻²) skin-friction. (-)
C_p	specific heat at constant pressure, $(J \text{ kg}^{-1} \text{ K}^{-1})$
Da D	Darcy number, (-) Brownian diffusion coefficient (-)
D_B D_u	Dufour number, (-)
E_c	Eckert number, (-)
G_r G_c	Grashof number, (-) modified Grashof number, (-)
K'	permeability of the porous medium, (-)
Kr k	chemical reaction parameter, (-)
L_{ρ}	Lewis number, (-)
N _b	Brownian parameter, (-)
N _t	thermophoresis parameter, (-)
N _u	local Nusselt number, (-)
P_r	Prandtl number, (-)
q_r	unidirectional radiative heat flux. (kg m ^{-2})
Q_{l}^{*}	radiation absorption, (-)
Q_{\circ}	heat absorption quantity, (-)
S_h	Sherwood number, (-)
T T.,	Temperature at the plate surface, (K)
w	ambient temperature as y tends to infinity, (K)
U_{\circ}	uniform velocity
и, v	velocity components
x, y Greek symbols	Cartesian co-ordinates
ρ	density of the fluid $(kg m^{-3})$
, И	dynamic viscosities
v	kinematic viscosity $(m^2 s^{-1})$
β	thermal expansion co-efficient
β^*	concentration expansion co-efficient
κ	thermal conductivity, $(Wm^{-1} K^{-1})$
σ_s	Stefan-Boltzmann constant, 5.6697×10^{-8} (W/m ² K ⁴)
Abbreviations	. ,
EFDM	explicit finite difference method
ODE	ordinary differential equation
PDE	partial differential equations
SCA	stability and convergence analysis

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APPENDIX A

To obtain the steady-state solutions of the computation, the calculation has been carried out up to non-dimensional time, $\tau = 10$ to 50. The velocity, temperature and concentration profiles shows a minor change after non-dimensional time, $\tau = 30$ (Fig. A1). Therefore, the solution for $\tau \ge 30$ is considered as steady-state solution.



Fig. A1. Illustration of velocity profiles for different dimensionless time steps, $\tau = 10$ to 50, where, $G_r = 10.0$, $G_c = 5.0$, $D_u = 0.03$, $Q_1 =$ 0.06, $N_v = 0.003$, $K_r = 0.5$, P = 2.0, $P_r = 1.38$, $E_c = 0.01$, $L_e = 1.0$, R = $0.05, M = 0.30, N_t = 0.10, D_a = 1.00, N_b = 0.10.$